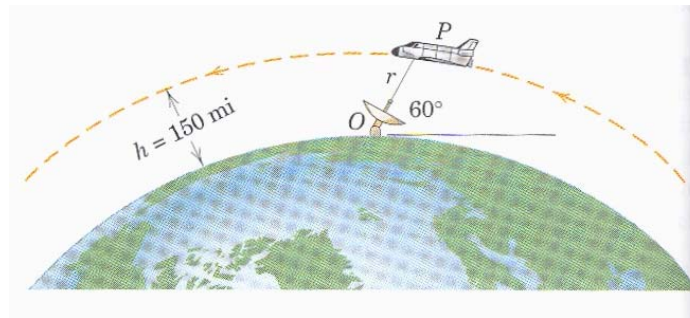


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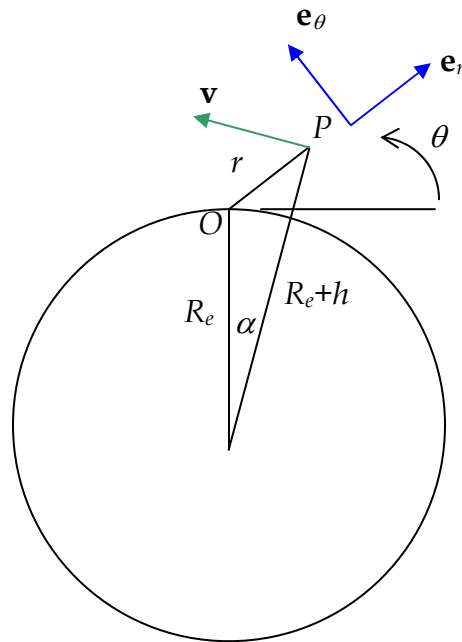
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Example 6.2

At the instant depicted in the figure, the radar station at O measures the range rate of the space shuttle P to be $\dot{r} = -12,272$ ft/s, with O considered fixed. If it is known that the shuttle is in a circular orbit at an altitude $h = 150$ mi, determine the orbital speed of the shuttle from this information.

**Solution:**

To see the geometry of this problem consider the following figure where P locates the position of the shuttle and O locates the origin of the radar dish:



In the figure above R_e is the radius of the earth and is given by $R_e = 3959$ mi. The data given is $h = 150$ mi and $\theta = 60^\circ$. From the given data it is also clear that:

$$r = \frac{h}{\sin \theta} = \frac{150}{\sin 60^\circ} = 173.2 \text{ mi}$$

From this value the angle α can be calculated using the cosine rule:

$$r^2 = R_e^2 + (R_e + h)^2 - 2R_e(R_e + h)\cos \alpha$$

or

$$\alpha = \arccos \left[\frac{R_e^2 + (R_e + h)^2 - r^2}{2R_e(R_e + h)} \right] = 1.23^\circ$$

Using polar coordinates the velocity of P is given by:

$$\mathbf{v} = \dot{r}\mathbf{e}_r + r\dot{\theta}\mathbf{e}_\theta$$

We only know the magnitude of the radial component ($v_r = \dot{r}$) of this velocity. We also know that the velocity must be tangent to the path of P (as shown by the green vector in the figure). Thus \mathbf{v} is perpendicular to the line from the center of the earth to P . Thus we know the angle between the unit vector \mathbf{e}_r and \mathbf{v} , say β . From the figure above the angle β can be calculated as:

$$\beta = 180 - (\alpha + \theta)$$

By taking the projection of \mathbf{v} on the r direction we obtain:

$$v_r = v \cos \beta$$

where v is the magnitude of \mathbf{v} . Thus:

$$v = \frac{v_r}{\cos \beta} = \frac{\dot{r}}{\cos \beta} = \frac{-12272}{\cos(180 - 1.23 - 60)} = 25498 \text{ ft/s}$$